

Amendments to the Claims:

The following listing of claims will replace all prior versions, and listings, of claims in the application:

1. (Currently Amended) A combustor for a gas turbine ~~engine comprising~~
engine, comprising:
 - a combustion chamber wall having formed therein at least one hole for admitting air into the combustion chamber;
 - at least one air intake chute aligned with said hole;
 - during operation a hoop stress field having regions of high and low stress concentration around said hole;
 - wherein said chute is ~~attached-provided with a flange disposed around one end thereof, and at least one tab that projects from the outer edge of said flange and is welded to~~ the combustor wall in a region of low stress ~~concentration~~. ~~concentration along the outer edge of the at least one tab such that the at least one tab provides a guide for the location and length of the weld.~~
2. (Original) A combustor as claimed in claim 1 wherein the chute is attached to the combustor wall in at least two regions of low stress concentration.
3. (Original) A combustor as claimed in claim 2 wherein areas where the chute is attached to the combustor wall are substantially in the same radial plane.
4. (Original) A combustor as claimed in claim 2 wherein the areas of attachment are provided on diametrically opposite sides of said chute.
5. (Canceled)
6. (Original) A combustor as claimed in claim 5 wherein the flange is circular.
- 7-8. (Canceled)

9. (Original) A combustor as claimed in claim 7 wherein the at least one tab projects from the edge of the flange up to about $0.14 \times$ (flange diameter).

10. (Original) A combustor as claimed in claim 7 wherein the at least one tab has a length of up to about $0.25 \times$ (flange diameter) of the diameter of the flange.

11-15. (Cancelled)